

Dedications

*This thesis is dedicated to all people
who supported me*

My mother

My father

My wife

*And lastly to my Grandmother, who
taught me to get up after a fall and
start again.*

Acknowledgment

This thesis is the result of work which I have been accompanied and supported by many people. First of all, I would like to thank Allah, who gives us everything.

Secondly, I would like to thank my supervisor, Dr. Moutaman Mirghani . Before being my thesis supervisor, he was one of my lecturers in Karary University. As a lecturer, he taught me all the required materials to achieve successfully this project. As a supervisor, although his time schedule was very busy, he has made the effort to be as available as possible to solve my problems. And thanks to his strong ability to listen to others, his answers were always consistent with my queries.

I would also like to express my deepest gratitude to Dr. Alkhawad Ali Alfaki, Dr. Mohamed Alhadi and ustaz Abdalla Saleh. Their advices and encouraging works kept my momentum going and my hope as strong as it could, and for all their help and guidance during this project.

Finally, I feel very grateful to my parents, my wife and my friends, who have always supported me, which perhaps is the greatest reason for my success.

مستخلص

المروحية الرباعية طائرة بدون طيار ذات اربعة دوّارات ثابتة، حيث يوجد دوارين إثنين علي كل محور ويتعامد كل من المحورين. تدور الدوّارات بواسطة أربعة محركات كهربائية لرفع جسم الطائرة في الهواء وتعتبر من الطائرات العمودية والتي تمتلك امكانية الاقلاع و الهبوط راسياً .

إن الهدف الرئيسي من هذا البحث هو دراسة وتطوير منظومة المروحية الرباعية لتكون قادرة علي الطيران بتحكم آلي، وتكون نواة للبحث والتطوير المستقبلي . ولتحقيق تلك الاهداف قد تضمن البحث دراسة شاملة للديناميكية الهوائية وعلي وجه الخصوص التركيز علي المروحيات وبالتحديد اسطح التحكم للمروحية ومكوناتها ، كما تم التطرق ايضاً الي عدد من انواع الطائرات ذات الاجنحة الدوّارة.

تم عمل تحليل للنموذج الرياضي للمروحية الرباعية وتم حساب القوي والعزوم المؤثرة عليها، ثم تم إستخراج معادلات الحركة التي تصفها. تم استخدام برنامج Matlab/Simulink بنظام التحكم في الحلقة المفتوحة للتحليق والانحدار والدوران و الانعراج للتحقق من استجابة النموذج. و قد كانت النتائج من المحاكاة التي تمت علي النموذج جيدة.

تم تصميم دائرة التحكم الالكترونية للمروحية ،بعد دراسة المكونات الالكترونية المستخدمة في بناء المنظومة مثل محركات التيار المستمر والمراوح والحساسات والبطاريات والمتحكم الدقيق. تم إستخدام برنامج XcopterCalc لتحليل المكونات الالكترونية المكونة للنظام ، وقد اعطي النظام مدي طيران وكفاءة جيدين عند استخدام دافع بين المتحكم والدوار.

تم إستخدام لغة برمجة micro C لكتابة اوامر التحكم في نظام المروحية، حيث يستقبل المتحكم الدقيق أوامر التحكم من القنوات الاربعية وتغذية العكسية للمقارنة مع المتحسسات وبذا يتم توليد إشارات التحكم للمحركات الاربعية . يتم التحكم في سرعة المحركات بواسطة التحكم بعرض النبضة للإشارة التي يتم ارسالها الي كل محرك علي حدا. كما تم إجراء محاكاة للدائرة التي تم تصميمها بواسطة برنامج Proteus .

Abstract

An autonomous quad-rotor is an unmanned aerial vehicle that has four-fixed rotors, where two rotors per axis and each of the axes are aligned with the other. Rotors are powered by four motors to lift and propel the aircraft. It is a type of a helicopter aircraft that has vertical take-off and landing capability.

The main objective of this research is to study and develop an autonomous quad-copter that is capable of flying under the control of an autopilot that's sustainable and expandable for future researches. In order to achieve that objective, this research encloses a comprehensive study of aerodynamics; focusing on helicopters in particular especially in the field of control surfaces and components, in addition to rapid survey to the history of helicopter development, as well as exploring some types of rotor-wing craft.

In addition to that, the analysis of Quad-Copter's mathematical model has been carried out; and the applied forces and moment affecting the Quad-copter has been calculated. Matlab/Simulink has been used to analyze the equations of motion that describes the mathematical model and to demonstrate the results of the open loop control system of the four movements, which are Hover, Pitch, Roll and Yaw. The results obtained from the simulation were acceptable and were as had been expected.

An electronic circuit is designed for the control of the Quad-Copter. Components used in the design and the implementation of the Quad-Copter such as DC motors, propellers, sensors, batteries and the microcontroller are all discussed in detail. The **XcopterCalc** program with $\pm 10\%$ accuracy was used to analysis for a particular combination of components, and found that the system has given good duration and efficiency when design include drive between controller and rotor. The Micro C programming language has been used for writing the controlling program for the system. The microcomputer receives the Hover, Pitch, Roll, and Yaw commands plus the feedback from the sensors and generates control signals to the four motors. The movement of the motors is controlled by varying PWM signals that are sent to each of the motors. The simulation of the designed circuit has been performed by using the **Proteus program**.

TABLE OF CONTENTS

Dedication	I
Acknowledgment	II
مستخلص	III
Abstract	IV
Table of Contents	V
List of Figures.....	XI
List of Tables	XIV

CHAPTER ONE

Introduction.....	1
1.1 Overview.....	1
1.2 Problem Statement.....	1
1.3 Objectives of the Research.....	2
1.4 Methodology	2
1.5 Layout of Thesis.....	2

CHAPTER TWO

Helicopter Theory of Flight and Control	4
2.1 Introduction.....	4
2.2 Early Helicopter History.....	4

2.2.1 First Successes.....	5
2.2.2 First VerticalFlight.....	6
2.2.3 New Developments.....	6
2.2.4 First True Helicopters.....	7
2.3 AerodynamicsofFlight.....	8
2.3.1 Forces Acting on the Aircraft.....	8
2.3.2 Airfoil	9
2.3.3 Airfoil Terminology and Definitions.....	9
2.3.4 Airfoil Types	10
2.3.5 Blade Twist.....	11
2.3.6 Lift and Drag Coefficient.....	11
2.3.7 Drag Coefficient (C_D).....	12
2.4 Introduction to Component of Helicopter.....	14
2.4.1 The Main Rotor System.....	14
2.4.1.1 Fully Articulated Rotor System.....	15
2.4.1.2 Semi Rigid Rotor System.....	15
2.4.1.3 Rigid Rotor System.....	16
2.4.1.4 Combinantion Rotor System.....	16
2.4. 2 Swash Plate Assembly.....	17
2.4. 3 Anti Torque System	17

2.4.3.1 Tail Rotor.....	17
2.4. 3.2 Fenestron.....	18
2.4.3.3 Notar.....	18
2.4. 3 Powerplant	18
2.4. 4 Flight Controls.....	18
2.4. 4.1 Collective Pitch and Throttle Control.....	19
2.4.4.2 Cyclic Pitch Control.....	19
2.4.4.3 Anti torque Pedals and Heading Control.....	19
2.5 Advanced Type of Rotor Craft.....	20
 CHAPTER THREE	
Quad Copter UAV.....	23
3.1 UAV Background.....	23
3.2 Modern UAV.....	24
3.3 First Helicopter Unmanned Aircraft Systems.....	25
3.4 History of Quad Copter.....	27
3.4.1 Oehmichen No.2.....	27
3.4.2 De Bothezat Quad Rotor.....	28
2.4.3 Convert Aving's Model A Quad-Rotor.....	29
3.4.4 Curtiss-Wright VZ-7	29
3.5 Quad Copter System Design.....	30

3.6 Quad Copter Description.....	31
CHAPTER FOUR	
Modeling and Simulation of Quad Copter.....	34
4.1 Equation of Motion	34
4.1.1 External Force and Moments ‘torque’	35
4.2 Euler Angle and Quad Copter Attitude.....	37
4.2.1 Linear Quantities Transformation.....	38
4.2.2 Angular Velocities Transformation.....	39
4.2.3 Force and Moment with Attitude Acting on Quad Copter...	41
4.3 General Moments and Forces Acting to Quad Copter.....	42
4.3.1 Simplify Equations of Motion.....	44
4.4 Simulation and Control by Simulink	47
4.4.1 Altitude/ Hover Simulation	48
4.4.2 Forward Simulation.....	50
4.4.3 Lateral Motion Simulation (Right-left).....	51
4.4.4 Yaw Simulation.....	52
CHAPTER FIVE	
Design Of Control System	53
5.1 Rotor.....	53
5.1.1 Motors.....	53

5.1.1.1 Brushless Motors.....	55
5.1.2 Propellers.....	55
5.1.3 Gears.....	56
5.2 Battery.....	56
5.2.1 Zippy Flight Battery.....	57
5.3 XcopterCalcProgram	58
5.4 Sensors.....	60
5.4.1 Accelerometer.....	60
5.4.2 Gyroscope.....	60
5.4.3 Inertia Measurement Unit.....	60
5.5 Electronic Speed Control	60
5.6 Microcontroller.....	61
5.7 Software Development General Concepts.....	62
5.8 Flight Mode in Quad Copter.....	64
5.8.1 Software Flow Chart Of The Quad Copter Program.....	65
 CHAPTER SIX	
Conclusion and Recommendation.....	68
6.1 Conclusion.....	68
6.2 Recommendation for further development	69
References.....	70

Appendix A:MATLAB /SIMULINK BLOCK DIOGRAM.....	72
Appendix B: MATLAB Code	75
B-1: MATLAB Code for Pitch control.....	75
B-2: MATLAB Code for Roll Control.....	76
Appendix C:.....	77
C-1: Schematic 0f IMU-500.....	77
C-2: GAUI Brush Motor.....	78
C-3: Motor Specifications.....	78
C-3: A Comparison between some Types of Battery.....	79
C-4 : Zippy Flight Max Battery.....	79
C-5: Zippy Battery Specification.....	79
Appendix D:The C Code.....	80
Appendix E:Circuit Schematic On Proteus Program.....	87

List of figure:

Figure 2-1: Chinese Top.....	4
Figure2-2: Leonardo da Vinci's Helicopter.....	5
Figure 2-3: Sir George Cayley's Helicopter.....	5
Figure 2-4: Cornu'sHelicopter	6
Figure 2-5: D'Ascanio's Helicopter	7
Figure 2-6: Sikorsky's VS-300	7
Figure2-7:Four Forces Acting on a Helicopter in Forward Flight	9
Figure2-8:AerodynamicTerms of an Airfoil.....	10
Figure 2-9 a:Pressure Distribution over an Airfoil.....	11
Figure 2-9 b: Lift \ Drag &Angle of Incidence or Attack in Airfoil.....	11
Figure2-10:The Total Drag Curve.....	13
Figure 2-11: The C_L and C_D versus α	14
Figure 2-12: Fully Articulated Main Rotor.....	15
Figure 2-13: Semi RigidRotor System.....	16
Figure 2-14:Combinantion Rotor System.....	16
Figure 2-15 :Swash plate.....	17
Figure 2-16: Tail Rotor Anti torque.....	17

Figure 2-17:Notar Anti Torque system.....	18
Figure 2-18:Tail Rotor Angle and Thrust in Relation to Pedal Position...	20
Figure 3-1:Breguet-Richet Gyroplane.....	27
Figure 3-2: Oeminchen No.2 Quad-Rotor	28
Figure 3-3: De Bothezats quad-Rotor Design	28
Figure 3-4: Convert a Wings Model A quad-rotor Design	29
Figure 3-5:Curtiss-Wright VZ-7.....	29
Figure 3-6: Quad Copter /Hover mode.....	31
Figure 3-7:Roll Movement.....	32
Figure 3-8: Pitch Movement.....	32
Figure 3-9: Yaw Movement.....	33
Figure 4-1: EulerAngle.....	37
Figure 4-2:Angular Rates Transformation	39
Figure 4-3: Quad Copter Force &Moment	43
Figure 4- 4:Modelin Simulink.....	48
Figure 4-5: Linear & Angular Motion in Hover Mode.....	49
Figure 4-6: Linear & Angular Motion (pitch).....	50

Figure 4-7: Linear& Angular Motion (roll).....	51
Figure 4-8: Linear & Angular Movement (Yaw).....	52
Figure 5-1:Brushless Motor Internal and Outer Coil.....	53
Figure 5-2: Brushed Motor Internal	54
Figure 5-3: Slow Flyer Composite Propeller.....	56
Figure 5.4: Motor Characteristic.....	58
Figure 5-5:Flight Mode of Quad Copter.....	64
Figure 5-6: Block Diagram Of Quad Copter Software Design Mode.....	65
Figure 5-7: Flow Chart of Quad Copter.....	67
Figure A-1: Linear Equation Circuit in Matlab/ Simulink.....	72
Figure A-2: Angular Equation of Model MATLAB/ Simulink.....	73
Figure A-3: The Force (Lift) of Four Rotor Model	74
Figure C-1: Schematic Of IMU-500.....	77
Figure C-2: GAUI Brushless Motor.....	78
Figure C-3:Zippy Flight Max Battery.....	79
Figure E-1 : Electronic Circuit Schematic.....	87

List of Tables

Table 2-1:Some Major Type of Rotor Craft.....	20
Table 3-1: Different Types of UAV's.....	24
Table 5-1: Remarks of Quad Parameter when Include Drive.....	59
Table 5-2: Remarks of Quad Parameter without Include Drive.....	59
Table C-1:Brushless Motor Specifications.....	78
Table C-2: Comparison of lithium polymer batteries with other batteries.	79
Table 5-3: Zippy Battery Specification	79